EASA
European Union Aviation Safety Agency

APPROVAL CERTIFICATE

EASA.21J.293

Pursuant to Regulations (EU) 2018/1139 and (EU) 748/2012 and subject to the conditions specified below, the Agency hereby certifies

Heli-One (Norway) AS

Flyplassveien 250
N-4055 Sola
Norway

as a DESIGN ORGANISATION

approved according to Part 21, Section A, Subpart J.

CONDITIONS :

1. The approval is limited to that specified in the enclosed Terms of Approval, and
2. This approval requires compliance with the procedures specified in the Design Organisation Handbook, reference Part 21 Subpart J Design Organisation Exposition Heli-One (Norway) AS, in the latest revision, and
3. This approval is valid whilst the approved Design Organisation remains in compliance with Part 21, Section A, Subpart J.
4. Subject to compliance with the foregoing conditions, this approval shall remain valid until surrendered or revoked.

For the **European Union Aviation Safety Agency,**
Date of issue: 1 June 2021

Francesco Maria Caridei
Design Organisations Section Manager

An agency of the European Union

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Terms of Approval

Design Organisation Approval Certificate

EASA.21J.293

1 Scope

This Design Organisation Approval is applicable for the scope defined in Annex A for design work with regard to the airworthiness, operational suitability and environmental characteristics of the products.

2 Privileges

a) (Reserved)

b) (Reserved)

c) The holder of this design organisation approval shall be entitled, within the scope of this terms of approval, and under the relevant procedures of the design assurance system:
   1. to classify changes to a type-certificate or to a supplemental type-certificate and repair designs as "major" or "minor";
   2. to approve minor changes to a type-certificate or to a supplemental type-certificate and minor repair designs;
   3. (Reserved);
   4. (Reserved);
   5. [not applicable];
   6. to approve for certain aircraft the flight conditions under which a permit to fly can be issued in accordance with point 21.A.710(a)(2), except for permits to fly to be issued for the purpose of point 21.A.701(a)(15);
   7. to issue a permit to fly in accordance with point 21.A.711(b) for an aircraft it has designed or modified, or for which it has approved, in accordance with point 21.A.263(c)(6), the flight conditions under which the permit to fly can be issued, and where the holder of this design organisation approval itself:
      (i) controls the configuration of the aircraft, and
      (ii) attests conformity with the design conditions approved for the flight;
   8. [not applicable];
   9. [not applicable].

3 Obligations

The holder of this design organisation approval shall, within the scope of this terms of approval:

a) maintain the handbook required under point 21.A.243 in conformity with the design assurance system;
b) ensure that this handbook or the relevant procedures included by cross-reference are used as a basic working document within the organisation;
c) determine that the design of products, or changes or repairs thereto comply with the applicable specifications and requirements and have no unsafe features;
d) provide the Agency with statements and associated documentation confirming compliance with point (c), except for approval processes carried out in accordance with point 21.A.263(c);
e) provide to the Agency data and information related to the actions required under point 21.A.3B;
f) under the privilege of paragraph 2(c)(6), determine the flight conditions under which a permit to fly can be issued;
g) under the privilege of paragraph 2(c)(7), establish compliance with points (b) and (e) of point 21.A.711 before issuing a permit to fly to an aircraft;
h) designate data and information issued under the authority of the approved design organisation within the scope of its terms of approval as established by the Agency with the following statement: "The technical content of this document is approved under the authority of the DOA ref. EASA. 21J.293".

Date of issue: 01/06/2021

Francesco Maria Caridei
Design Organisations Section Manager

Annex A

Scope of work

		TC	STC	major changes	minor changes	major repairs	minor repairs	flight conditions	permit to fly
Large and small aeroplane									
Avionics									
	All areas		■	■	■	■	■	■	■
Cabin									
	All areas		■	■	■	■	■	■	■
Electrical Systems									
	All areas		■	■	■	■	■	■	■
Environmental Control Systems									
	All areas		■	■	■	■	■	■	■
Flight									
	Flight characteristics		■	■	■	■	■	■	■
Hydro-Mechanical Systems									
	All areas		■	■	■	■	■	■	■
Powerplant and Fuel Systems									
	All areas		■	■	■	■	■	■	■
Structures									
	All areas		■	■	■	■	■	■	■

			TC	STC	major changes	minor changes	major repairs	minor repairs	flight conditions	permit to fly
Large and small rotorcraft										
	Avionics									
		All areas		■	■	■	■	■	■	■
	Cabin									
		All areas		■	■	■	■	■	■	■
	Electrical Systems									
		All areas		■	■	■	■	■	■	■
	Environmental Control Systems									
		All areas		■	■	■	■	■	■	■
	Flight									
		Flight characteristics		■	■	■	■	■	■	■
	Hydro-Mechanical Systems									
		All areas		■	■	■	■	■	■	■
	Powerplant and Fuel Systems									
		All areas		■	■	■	■	■	■	■
	Rotor Drive Systems									
		All areas		■	■	■	■	■	■	■
	Structures									
		All areas		■	■	■	■	■	■	■
Turbine engine										
	Propulsion									
		All areas						■		

Legend:

	Title for category of product
	Title for design scope
	Title for design area

■	Within scope
	Outside scope

List of products

[not applicable]

Limitations

	Limitations common to all products and activities
	Changes affecting OSD are excluded

Product	Limitations particular to each product
Large and small aeroplane	For non-TCH activity: Primary structure is excluded Pressurized structure is excluded
Turbine engine	For non-TCH activity: limited to rotorcraft turbine engines engine rotating parts are excluded